



University of California, San Diego

Faculty of Engineering

DESIGN OF BEAM COLUMNS

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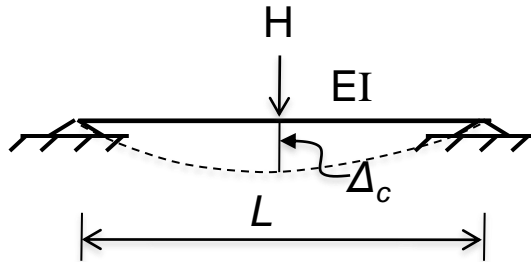
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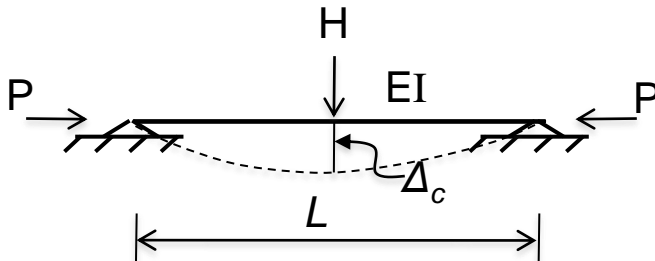
INTRODUCTION

Consider a simply supported beam with a midspan point load:



$$\Delta_c = \frac{HL^3}{48EI}$$

What if the beam is now subjected to axial compression?



$$\Delta_c > \frac{HL^3}{48EI}$$

The resulting moment will be amplified by the axial load:

$$M = \frac{HL}{4} + P\Delta_c, \quad \text{where} \quad \Delta_c > \frac{HL^3}{48EI}$$



INTRODUCTION

A member subjected to combined bending and axial load is called a beam column.

The amplification of the moment due to the axial load is called 2nd order effects.

$$M = \frac{HL}{4} + P\Delta_c$$

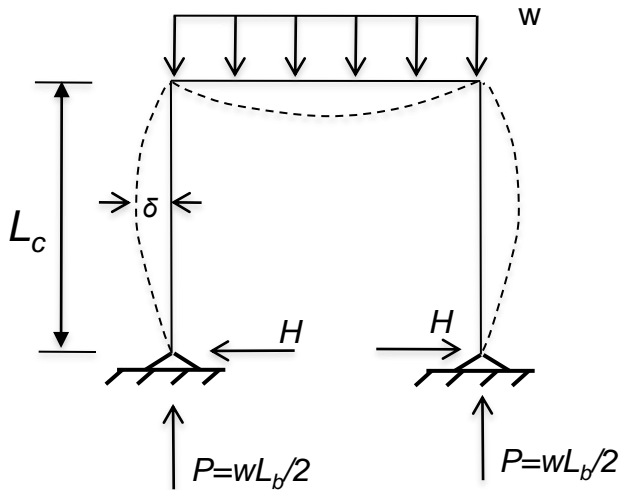
First order moment 2nd order moment

Engineers are encouraged to perform an exact 2nd order analyses to determine 2nd order effects. This is now getting much more possible with advanced software programs.

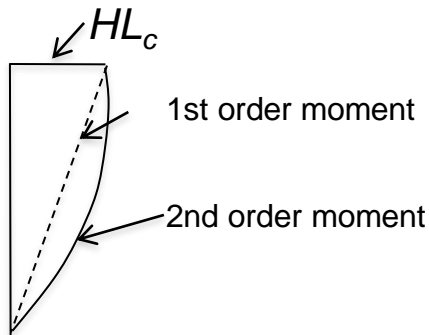
If a computer program is used to calculate 2nd order moments it is important to consider “both” types of moment amplifications.



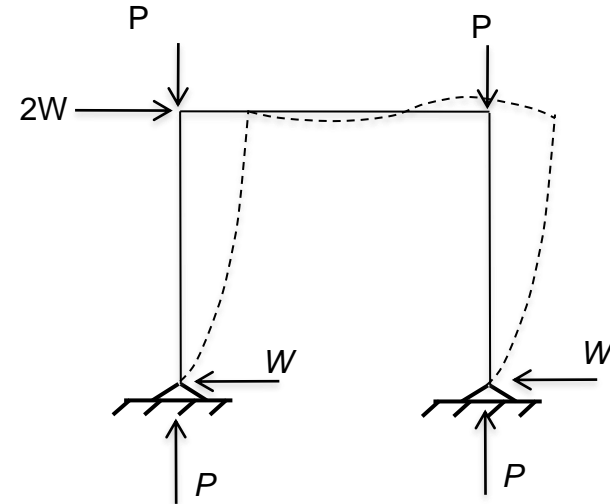
MOMENT AMPLIFICATION



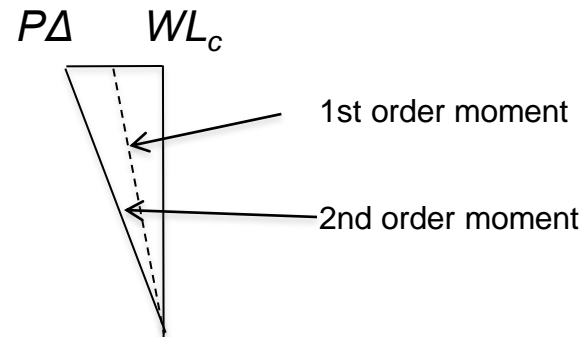
No Sway Moments: $P\delta$



Maximum moment can occur at the end or away from the end if $P\delta$ is large.



No Sway Moments: $P\Delta$



Maximum moment occurs at column ends.



MOMENT AMPLIFICATION

A Second order analysis considers both $P\delta$ and $P\Delta$.

The AISC Specification provides amplification factors which can be used with the moments from a 1st order analysis. The approximate 2nd order effects are covered in Chapter C Section C2.

$$M_r = B_1 M_{nt} + B_2 M_{lt}$$

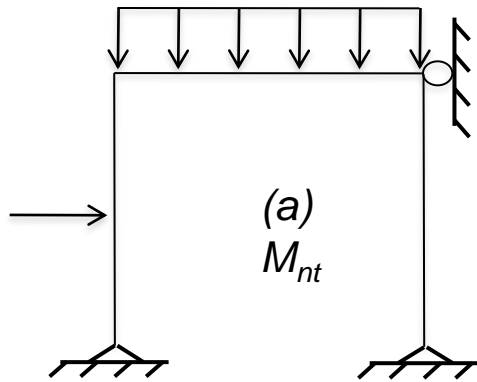
$$P_r = P_{nt} + B_2 P_{lt}$$

($P\delta$) M_{nt} is the required flexural strength in the member assuming there is no lateral translation of the frame.

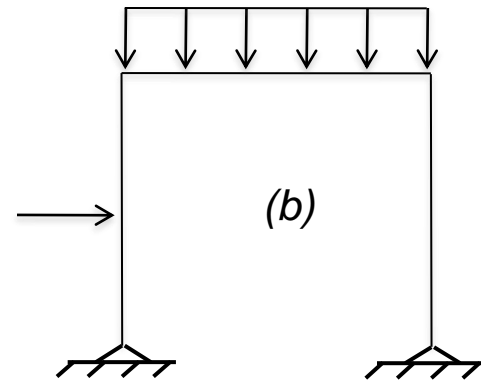
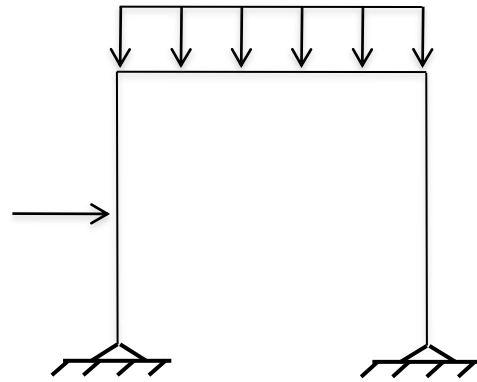
($P\Delta$) M_{lt} is the required flexural strength in the member as a result of the lateral translation of the frame only.



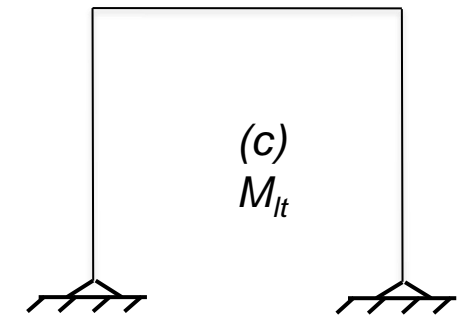
MOMENT AMPLIFICATION



Perform a 1st order analyses on the frame preventing lateral transformation. This gives M_{nt} .



Perform a 1st order analyses on the frame allowing lateral transformation.

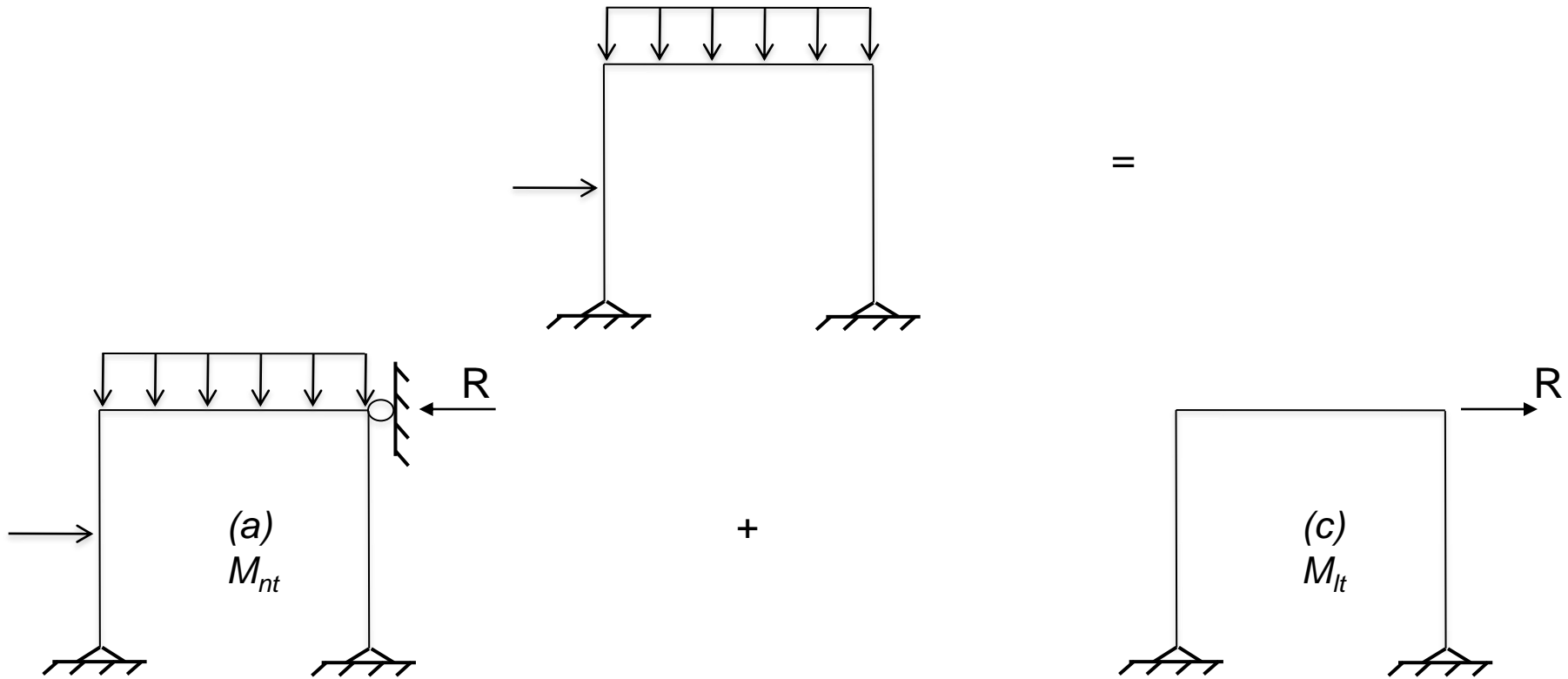


Subtract the moments from (a) from the moments in (b). This gives M_{lt} .

$$M_r = B_1 M_{nt} + B_2 M_{lt}$$



MOMENT AMPLIFICATION



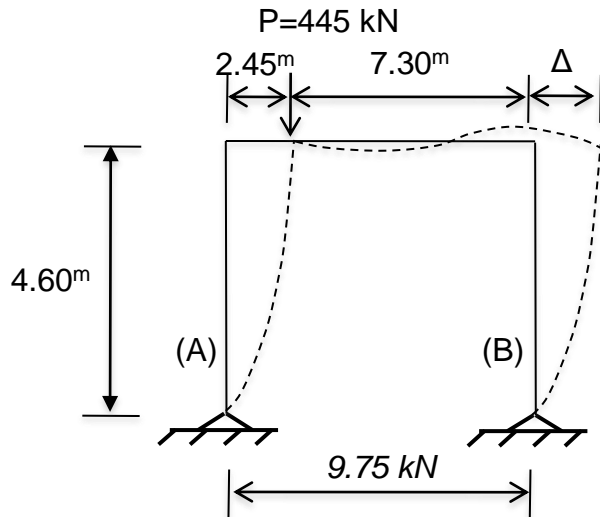
Perform a 1st order analyses on the frame preventing lateral transformation. This gives M_{nt} .

Compute M_{lt} directly.

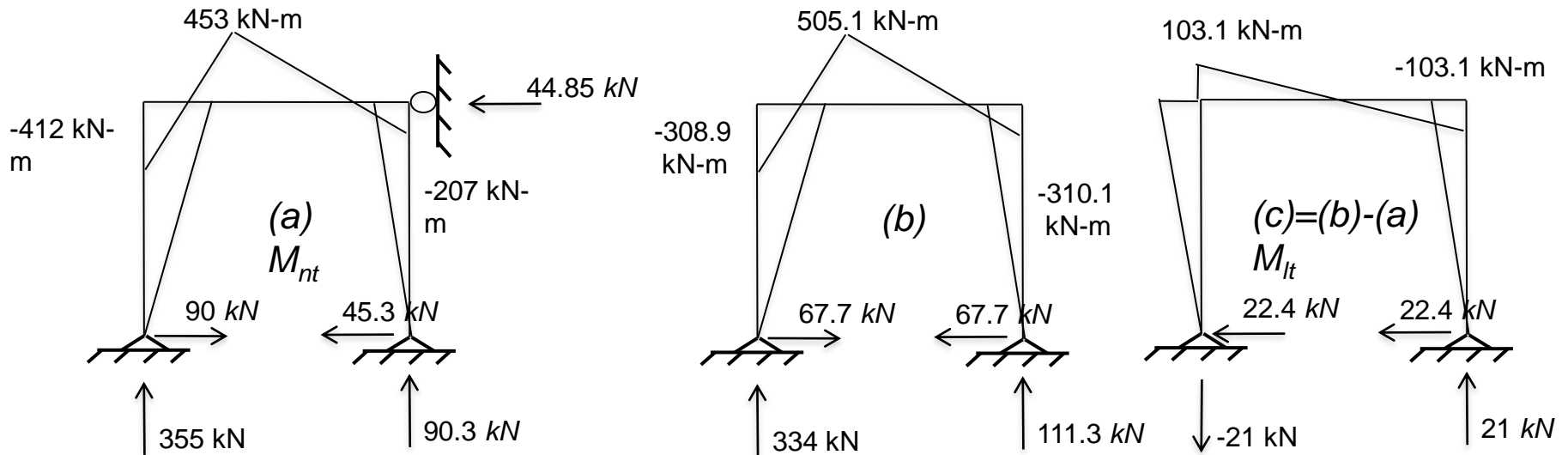
$$M_r = B_1 M_{nt} + B_2 M_{lt}$$



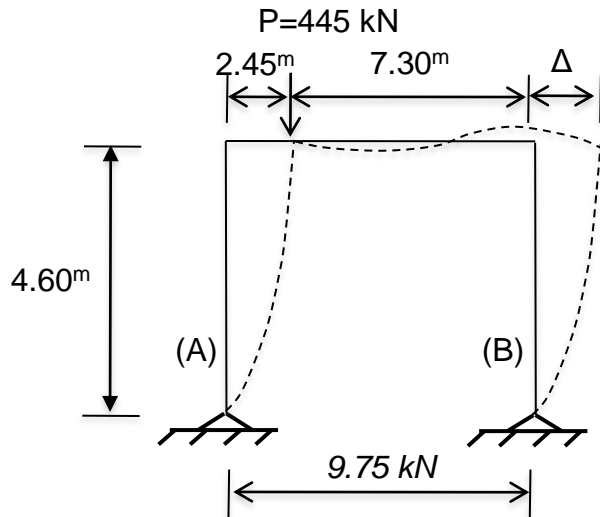
EXAMPLE



Columns and beams are: W460 × 82
 $I_x=370 \times 10^6 \text{ mm}^4$
 $r_x=189 \text{ mm}$
 $A=10400 \text{ mm}^2$
 $\Delta= 20\text{mm}$ from 1st order analyses



EXAMPLE



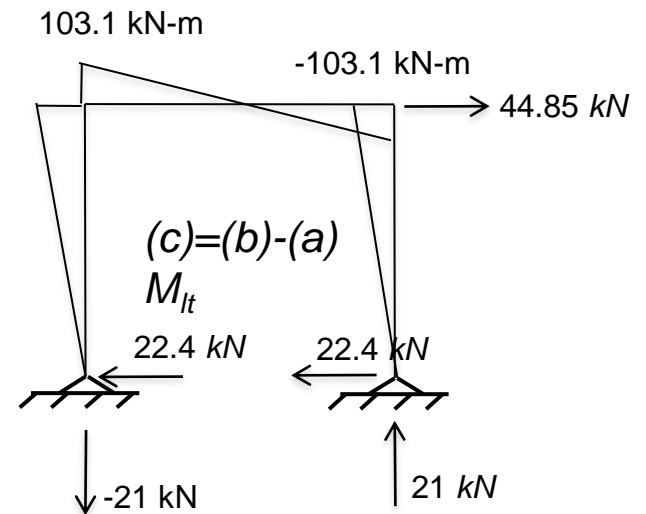
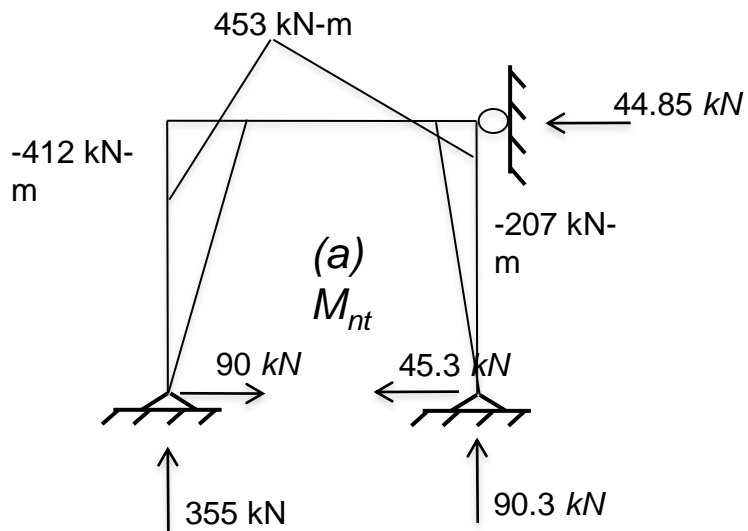
Columns and beams are: W460 × 82

$I_x=370 \times 10^6 \text{ mm}^4$

$r_x=189 \text{ mm}$

$A=10400 \text{ mm}^2$

$\Delta= 20 \text{ mm}$ from 1st order analyses



EXAMPLE

Calculate B_2 : P Δ Moment. $B_2 = \frac{1}{1 - \frac{\alpha P_{story}}{P_{estory}}} \geq 1 (A-8-6)$

Calculate P_{e2} for Column A and use it for both columns:
 $m=1$; $P/A=(445\text{kN}\times 7.30\text{m}/9.75\text{m})/10400\text{mm}^2=32\text{ MPa}$; $\tau=1.0$

$$G_T = \frac{1.0 \sum \left(\frac{I_x}{L_c} \right)_{W 460 \times 82} = \left(\frac{370 \times 10^6}{4.6^m} \right)}{1.0 \sum \left(\frac{I_x}{L_b} \right)_{W 460 \times 82} = \left(\frac{370 \times 10^6}{9.75^m} \right)} = 2.12 \rightarrow k_x = 2.7$$

$$G_B = \text{pinned} = \infty$$

$$P_{e2} = \frac{\pi^2 EI}{(kL)^2} = \frac{3.14^2 \times 200000 \text{ N/mm}^2 \times 370 \times 10^6 \text{ mm}^4}{(2.7 \times 4600 \text{ mm})^2} = 4842 \text{ kN}$$



EXAMPLE

Calculate B_2 : $P\Delta$ Moment.

$$B_2 = \frac{1}{1 - \frac{\alpha P_{story}}{P_{e story}}} = \frac{1}{1 - \frac{1.0 \times 445^{kN}}{4842^{kN} \times 2}} = 1.05$$



EXAMPLE

Calculate B_1 : P δ Moment.

Calculate P_{e1} for Column A and Column B with worst k factor, $k=1.0$:

$$P_{e1} = \frac{\pi^2 EI}{(kL)^2} = \frac{3.14^2 \times 200000 \text{ N/mm}^2 \times 370 \times 10^6 \text{ mm}^4}{(1.0 \times 4600 \text{ mm})^2} = 34516 \text{ kN}$$

Calculate C_m using Eq. C2-4, AISC-05.

$$C_m = 0.6 - 0.4(M_1 / M_2)$$

M_1/M_2 ratio is the ratio of the moments at each end of the column. M_1 is the smallest moment, M_2 is the larger moment. If the column is under double curvature M_1/M_2 is (+), if the column is under single curvature the ratio is (-).



EXAMPLE

Calculate C_m using Eq. C2-4, AISC-05.

$$C_m = 0.6 - 0.4(M_1 / M_2)$$

$$M_1 = 0 \Rightarrow C_m = 0.6$$

$$B_1 = \frac{0.6}{1 - 1.0P_r / P_{e1}} \geq 1.0$$



EXAMPLE

Calculate B_1 for Column A:

$$B_1 = \frac{0.6}{1 - 1.0x \frac{445kN \frac{7.3^m}{9.75^m}}{34516^{kN}}} = 0.606 \leq 1.0; \text{ therefore } B_1 = 1.0$$

Calculate B_1 for Column B:

$$B_1 = \frac{0.6}{1 - 1.0x \frac{445kN \frac{2.45^m}{9.75^m}}{34516^{kN}}} = 0.602 \leq 1.0; \text{ therefore } B_1 = 1.0$$



EXAMPLE

2nd order Moments:

$$\text{Column A} \Rightarrow M_u = 1.0 \times 412^{kN-m} + 1.05 \times (-103.1)^{kN-m} = 303.7 kN - m$$

$$\text{Column B} \Rightarrow M_u = 1.0 \times 207^{kN-m} + 1.05 \times (103.1)^{kN-m} = 315.3 kN - m$$

2nd order Axial Forces:

$$\text{Column A} \Rightarrow P_u = 355.1^{kN} + 1.05 \times (-21)^{kN} = 333.1 kN$$

$$\text{Column B} \Rightarrow P_u = 90.3^{kN} + 1.05 \times (21)^{kN} = 112.3 kN - m$$



AXIAL FORCE-MOMENT INTERACTION

Once the axial force and 2nd order moments are calculated, use the axial force-moment interaction formula to evaluate the capacity. These equations are in Chapter H on p. 16.1-70. Use the same equations for tension and compression.

$$(a) \text{ For } \frac{P_r}{P_c} \geq 0.2 \Rightarrow \frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

$$(b) \text{ For } \frac{P_r}{P_c} < 0.2 \Rightarrow \frac{P_r}{2P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

$M_c = \Phi_b M_n$ = available flexural strength

M_r = required flexural strength using LRFD load combinations

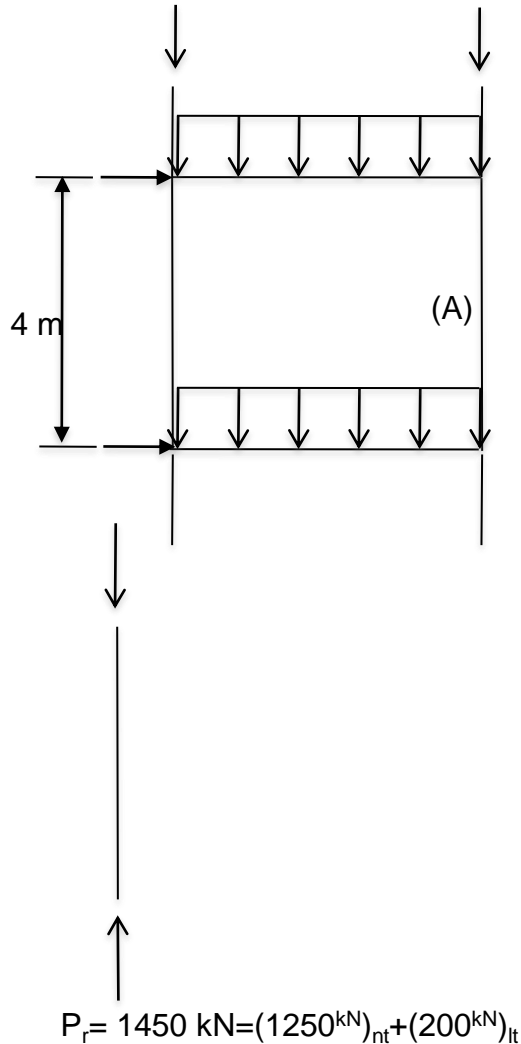
x = subscript relating to strong axis bending

y = subscript relating to weak axis bending

P_r = required axial compressive strength using LRFD load combinations

$P_c = \Phi_c P_n$ = design axial compressive strength

EXAMPLE

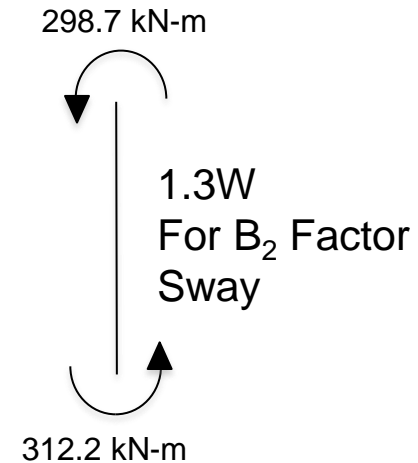
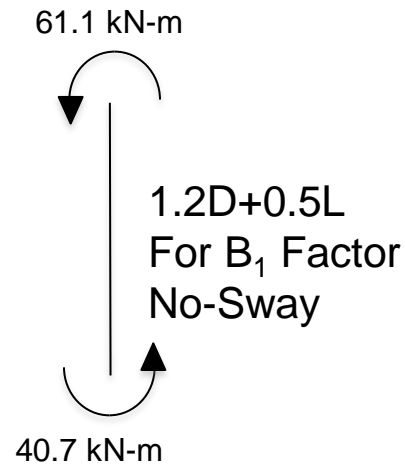


Consider a portion of a multistory frame.

Beams are: W410 × 149

$I_x = 619 \times 10^6 \text{ mm}^4$

Unbraced in plane and braced each story out-of-plane.



EXAMPLE

Try: W360 × 110

$$I_x = 331 \times 10^6 \text{ mm}^4$$

$$I_y = 55.7 \times 10^6 \text{ mm}^4$$

$$r_x = 154 \text{ mm}$$

$$r_y = 63.1 \text{ mm}$$

$$A = 14000 \text{ mm}^2$$

$$Z_x = 2060000 \text{ mm}^3$$

$$m = 1.0; P/A = 1450000 \text{ N} / 14000 \text{ mm}^2 = 103.6 \text{ MPa}; \tau = 1.0 \text{ (Table-3.1)}$$

In-Plane: Sway

$$G_T = G_B = \frac{1.0 \sum \left(\frac{I_x}{L_c} \right)_{W 360 \times 110}}{1.0 \sum \left(\frac{I_x}{L_b} \right)_{W 410 \times 149}} = \frac{2x \left(\frac{331 \times 10^6}{4.0^m} \right)}{\left(\frac{6190 \times 10^6}{6.0^m} \right)} = 1.6 \Rightarrow k_x = 1.5$$

Out-of-Plane: No Sway

$$k_y = 1.0$$

$$\frac{k_y L}{r_y} = \frac{1.0 \times 4000^{mm}}{63.1^{mm}} = 63.4$$

$$\frac{k_x L}{r_x} = \frac{1.5 \times 4000^{mm}}{154^{mm}} = 39.0$$

Controls



EXAMPLE

Use Column Load Tables with $kL_y=4$ m. p 3-21; $\Phi P_n=3060$ kN.

$$\frac{P_r}{P_c} = \frac{P_u}{\phi P_n} = \frac{1450^{kN}}{3060^{kN}} = 0.47 \geq 0.2 \Rightarrow \text{Therefore use Eq. H1-1a}$$

$$\frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) = \frac{P_u}{\phi P_n} + \frac{8}{9} \frac{M_u}{\phi M_n} \leq 1.0$$

$$M_r = M_u = B_1 M_{nt} + B_2 M_{lt}$$

Calculate B_1 :

$$B_1 = \frac{C_m}{1 - \alpha P_r / P_{e1}} \geq 1.0$$

Double curvature (+)

$$C_m = 0.6 - 0.4(M_1 / M_2)$$

$$C_m = 0.6 - 0.4(40.7^{kN-m} / 61.1^{kN-m}) = 0.33$$



EXAMPLE

$$P_{e1} = \frac{\pi^2 EI}{(kL)^2} = \frac{3.14^2 \times 200000 \text{ N/mm}^2 \times 331 \times 10^6 \text{ mm}^4}{(1.0 \times 4000 \text{ mm})^2} = 40836 \text{ kN}$$

$$B_1 = \frac{C_m}{1 - \alpha P_r / P_{e1}} = \frac{0.33}{1 - 1.0 \frac{1450 \text{ kN}}{40836 \text{ kN}}} = 0.342 \leq 1.0 \Rightarrow B_1 = 1.0$$

Calculate B_2 :

$$B_2 = \frac{1}{1 - \frac{\alpha \sum P_{nt}}{\sum P_{e2}}}$$



EXAMPLE

$$P_{e2} = \frac{\pi^2 EI}{(kL)^2} = \frac{3.14^2 \times 200000 \text{ N/mm}^2 \times 331 \times 10^6 \text{ mm}^4}{(1.5 \times 4000 \text{ mm})^2} = 18149 \text{ kN}$$

$$B_2 = \frac{1}{1 - \frac{\alpha \sum P_{nt}}{\sum P_{e2}}} = \frac{1}{1 - 1.0 \frac{2 \times 1450 \text{ kN}}{2 \times 18149 \text{ kN}}} = 1.074$$

$$(M_u)_{top} = 1.0 \times 61.1 \text{ kN-m} + 1.074 \times (298.7) \text{ kN-m} = 381.9 \text{ kN-m}$$

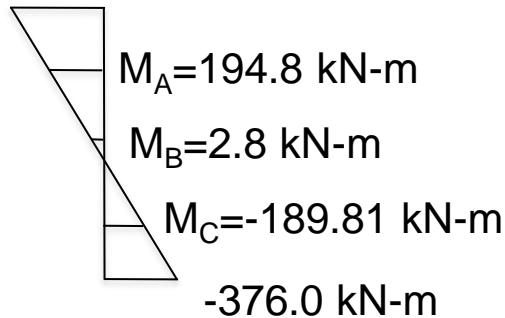
$$(M_u)_{bottom} = 1.0 \times 40.7 \text{ kN-m} + 1.074 \times (312.2) \text{ kN-m} = 376.0 \text{ kN-m}$$



EXAMPLE

Calculate C_b factor for the column

381.9 kN-m



$$C_b = \frac{12.5 M_{\max}}{2.5 M_{\max} + 3 M_A + 4 M_B + 3 M_C}$$

$$C_b = \frac{12.5 \times 382^{kN-m}}{2.5 \times 382^{kN-m} + 3 \times 194.8^{kN-m} + 4 \times 2.85^{kN-m} + 3 \times 189.1^{kN-m}} = 2.26$$



EXAMPLE

W360 × 110: $L_r=8.47$ m $>L_b=4$ m; therefore check $C_b\phi M_r$

$$\phi M_r = 457 \text{ kN} - m; \phi M_p = 640 \text{ kN} - m$$

$$C_b\phi M_r = 2.26 \times 457 \text{ kN} - m = 1033 \text{ kN} - m \rangle \phi M_p = 640 \text{ kN} - m$$

$$\phi M_n = \phi M_p = 640 \text{ kN} - m$$

Calculate $P_r=P_{nt}+B_2P_{lt}$

$$P_r = P_{nt} + B_2P_{lt} = 1.0 \times 1250^{kN} + 1.074 \times 200^{kN} = 1465 \text{ kN}$$



EXAMPLE

Use the interaction equation to see if the beam-column W360 × 110 is adequate or not.

$$\frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) = \frac{P_u}{\phi P_n} + \frac{8}{9} \frac{M_u}{\phi M_n} \leq 1.0$$

$$\frac{1465^{kN}}{3060^{kN}} + \frac{8}{9} \frac{382^{kN-m}}{640^{kN-m}} = 1.00 \leq 1.0 \text{ OK}$$

Use W360 × 110. If W360 × 110 was inadequate use a bigger beam size



DESIGN

We have discussed the design of beam columns using the interaction equations given in Chapter H.

$$(a) \text{ For } \frac{P_r}{P_c} \geq 0.2 \Rightarrow \frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

$$(b) \text{ For } \frac{P_r}{P_c} > 0.2 \Rightarrow \frac{P_r}{2P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

The P_r/P_c or $P_u/\Phi P_n$ ratio that is used is the larger ratio for strong axis or weak axis buckling.

The examples that we have considered so far have been for symmetrical loading. If we have unsymmetrical loading we can use the ΣP concept $(\Sigma P_u/\Phi)\Sigma P_n$.

It is important that we check a column for buckling in both directions and use the larger ratio of P_r/P_c .



DESIGN

Trial Size

We can get a trial size to use by calculating an “equivalent” load for the member:

If the member behaves more like a column:

$$P_{uEQ} = P_u + M_x(2/d) + M_y(7.5/b_f) \quad \text{Use column load tables.}$$

If the member behaves more like a beam:

$$M_{uEQ} = P_u d/2 + M_x + M_y \quad \text{Use Z tables.}$$

This is a judgement call in which you look at the relative magnitudes of the column contribution from axial force and moment.

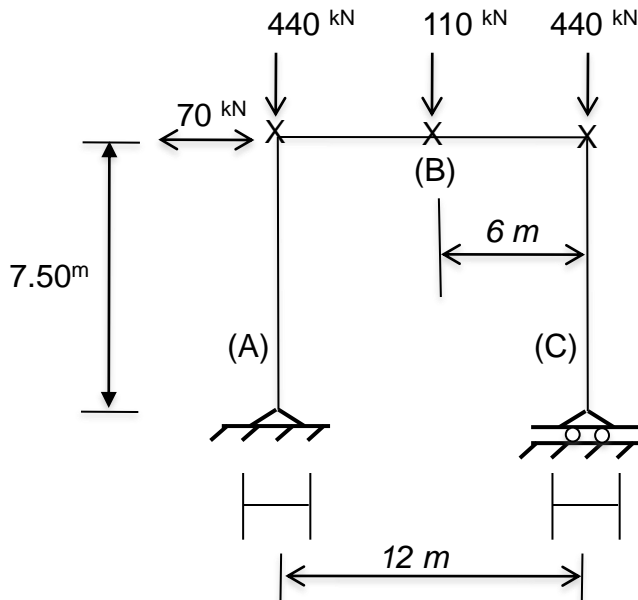
ie. If $M_x(2/d)$ is $\gg P_u$, it would be obviously be more efficient to select a deeper “beam” type section than going with a stocky “column” type section.



EXAMPLE

Design Column A and C and Beam B

$$\underline{F_y = 345 \text{ MPa}}$$



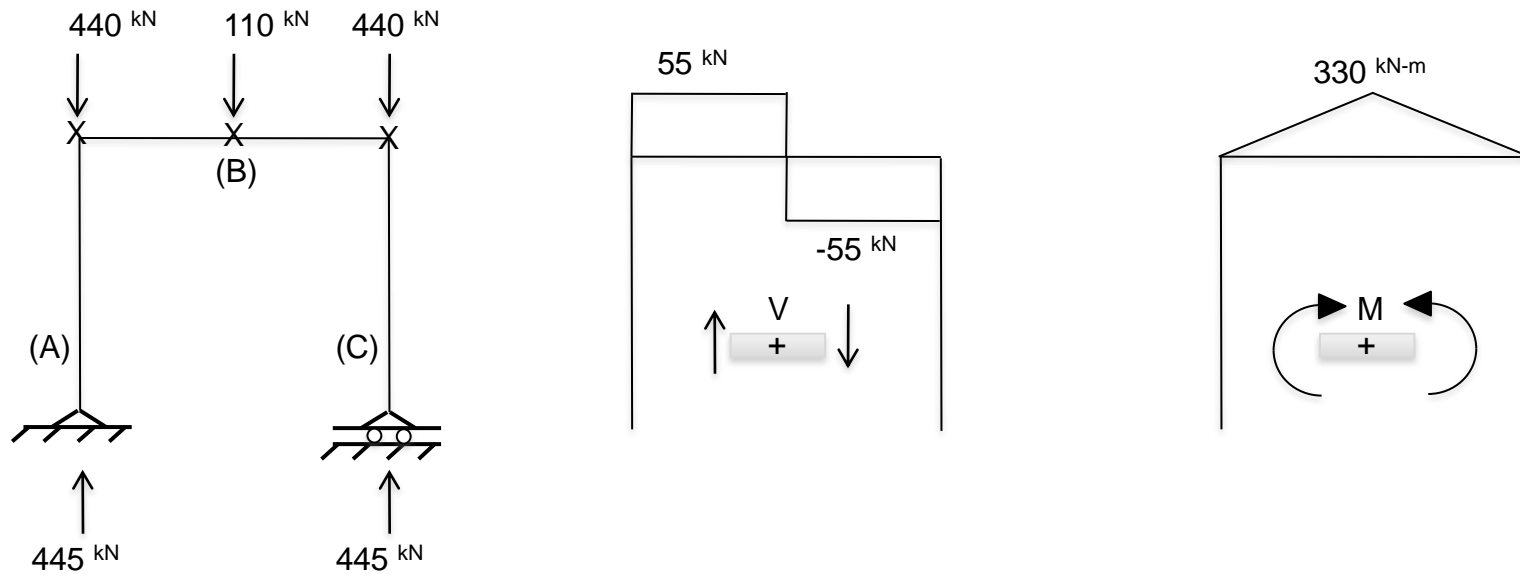
Columns are braced at top and bottom out of plane. Factored loads are shown. Use same size members for columns A and C.

x: brace locations



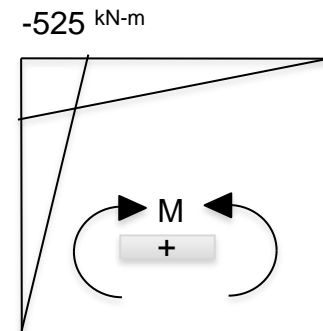
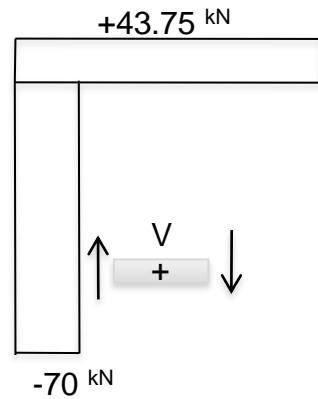
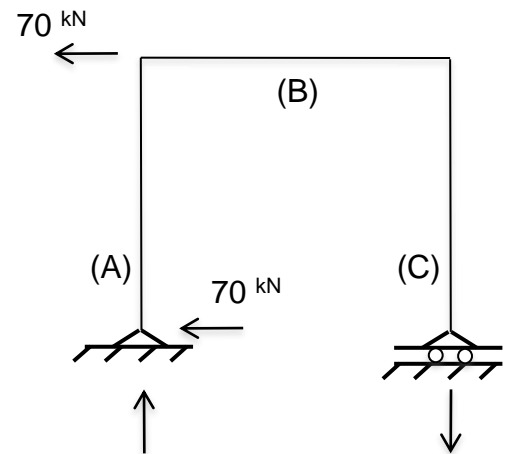
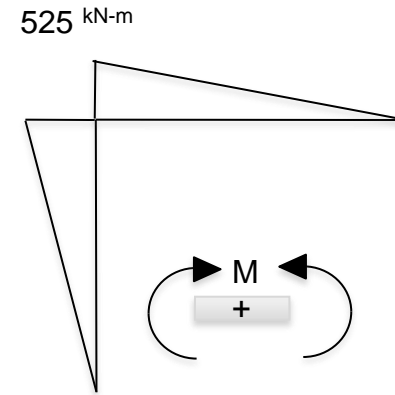
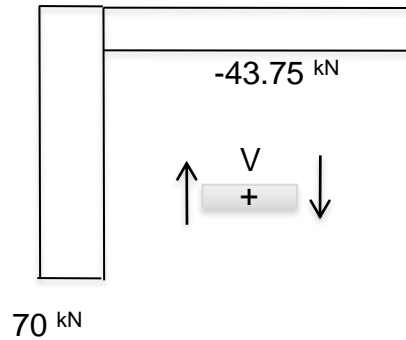
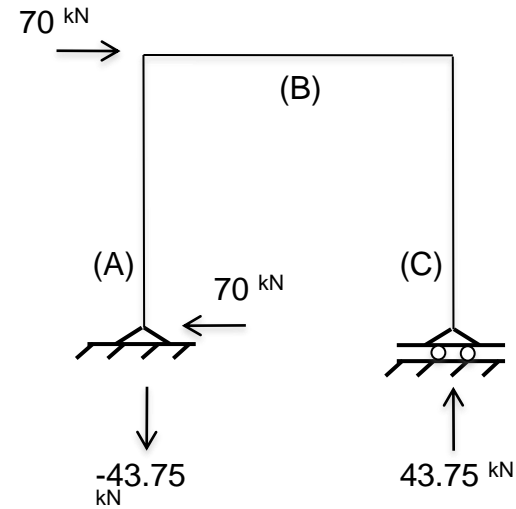
EXAMPLE

Shear and Bending Moment diagrams under gravity loads:



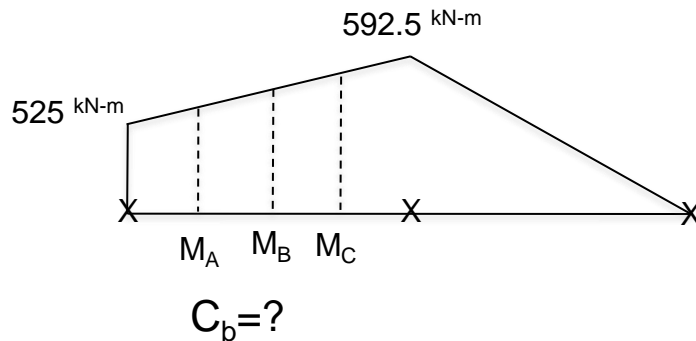
EXAMPLE

Shear and Bending Moment diagrams under wind loads:



EXAMPLE

Design the beam.



$$M_A = 541.9 \text{ kN-m}$$

$$M_B = 558.8 \text{ kN-m}$$

$$M_C = 575.6 \text{ kN-m}$$

$$C_b = \frac{12.5 M_{\max}}{2.5 M_{\max} + 3 M_A + 4 M_B + 3 M_C}$$

$$C_b = \frac{12.5 \times 592.5 \text{ kN-m}}{2.5 \times 592.5 \text{ kN-m} + 3 \times 541.9 \text{ kN-m} + 4 \times 558.8 \text{ kN-m} + 3 \times 575.6 \text{ kN-m}} = 1.05$$



EXAMPLE

Design the beam.

$$M_u = 592.5 \text{ kN} - m$$

$$L_b = 6 \text{ m}$$

Try a W610 × 82 (Z-Tables p.4-19)

$$\phi_b M_p = 683 \text{ kN} - m$$

$$L_r = 3.91 \text{ m}$$

$$L_p = 1.45 \text{ m}$$

LTB Check: $L_b \gg L_r$, this section clearly wont work.

Try a W530 × 92 (Z-Tables p.4-19)

$$\phi_b M_p = 733 \text{ kN} - m$$

$$L_r = 5.07 \text{ m}$$

$$L_p = 1.90 \text{ m}$$

$$\phi_b M_r = 514 \text{ kN} - m$$

LTB Check: p. 4-154 or 4-156; $\Phi M_o = 300 \text{ kN-m}$; Wont work



EXAMPLE

Design the beam.

$$M_u = 592.5 \text{ kN} - m$$

$$L_b = 6 \text{ m}$$

Try a W610 × 101 (Z-Tables p.4-19)

$$\phi_b M_p = 900 \text{ kN} - m$$

$$L_r = 5.33 \text{ m}$$

$$L_p = 2.02 \text{ m}$$

Try a W610 × 113 (Z-Tables p.4-19)

$$\phi_b M_p = 1020 \text{ kN} - m$$

$$L_r = 5.51 \text{ m}$$

$$L_p = 2.07 \text{ m}$$

$$\phi_b M_r = 715 \text{ kN} - m$$

LTB Check: p. 4-154; $\Phi M_o = 602$; $C_b \Phi M_o = 632 \text{ kN} - m$.

USE W610 × 113



EXAMPLE

Design the beam.

Shear should not be a problem but let us check it anyway.

$$V_u = 98.75 \text{ kN} - m$$

$$\frac{h}{t_w} = 48.9 \leq 2.45 \sqrt{\frac{E}{F_{yw}}} \quad L_p = 2.02 \text{ m}$$

$$\phi V_n = 0.9 \times 0.6 \times A_w F_{yw} = 0.9 \times 0.6 \times 608^{mm} \times 11.2^{mm} \times 345^{MPa} = 1268 \text{ kN}$$

$$\phi V_n = 1268 \text{ kN} \geq V_u = 98.75 \text{ kN}$$

Displacements should also be checked. However, since live loads are not given we will assume deflections are OK.



EXAMPLE

Design the columns.

Column A is more critical than column C. Same P_u but column A has also bending.

Does the member behave more like a column or a beam?

$$P_{uEQ} = P_u + M_x (2/d) + M_y (7.5/b_f)$$

Use average.

$$P_{uEQ} = \frac{990^{kN}}{2} + 525^{kN-m} \frac{2}{0.31^m} = 495^{kN} + 3387^{kN}$$

Assume a W310 is used.

Bending dominates. Choose a section that is more efficient in bending. Obtain an equivalent moment in the beam column.



EXAMPLE

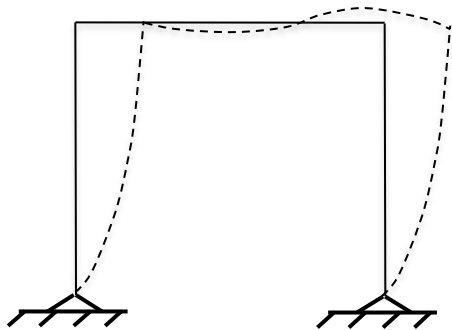
Design the columns.

Assume a W460 is used.

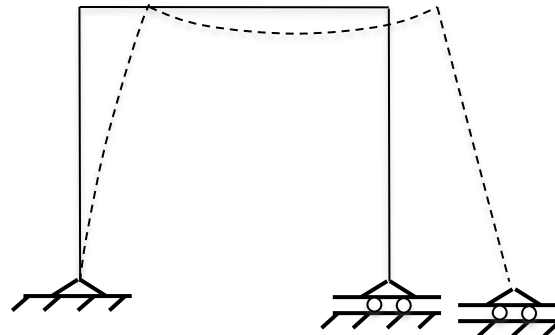
$$M_{uEQ} = \frac{P_u d}{2} + M_u = \frac{(995^{kN} / 2) \times 0.46^m}{2} + 525^{kN-m}$$

$$M_{uEQ} = 639.5 \text{ kN} - m$$

Look at possible “beam” type sections with $\Phi M_{p,req'd} = 640 \text{ MPa}$.
However, before spending a lot of time checking a particular section for this beam-column, consider the stiffness of beam (B) that provides the restraint.



Assumed stiffness: $6EI/L$.



Actual stiffness: $3EI/L$.

$$m = \frac{2EI/L}{6EI/L} = 0.33$$

EXAMPLE

Design the columns.

The M_{EQ} that is calculated does not reflect the low beam stiffness that will affect ΦP_n . However, we can use this knowledge to slightly increase M_{EQ} as we select a trial member size.

$$\phi M_p = 639.5 \text{ kN} - m \Rightarrow \text{Check Z - Tables}$$

$$W 610 \times 82 - \phi M_p = 683 \text{ kN} - m$$

$$W 530 \times 92 - \phi M_p = 733 \text{ kN} - m$$

Try

W530 x 92.

$$\phi M_p = 683 \text{ kN} - m \quad I_y = 23.8 \times 10^6 \text{ mm}^4 \quad I_x = 552 \times 10^6 \text{ mm}^4$$

$$A = 11800 \text{ mm}^2 \quad r_x = 216 \text{ mm} \quad r_y = 44.9 \text{ mm}$$



EXAMPLE

Design the columns.

Calculate the available column strength.

No Sway Mode: $kL_y=1.0$; kL_x can be assumed to be 1.0 due to low stiffness of the beam.

$$kL_y = 7500 \text{ mm} \Rightarrow \frac{kL_y}{r_y} = \frac{7500^{\text{mm}}}{44.9^{\text{mm}}} = 167$$

Table 3-345 p.6-148.

$$\frac{kL_y}{r_y} = 167 \Rightarrow \phi F_{cr} = 52.8 \text{ MPa}$$

$$\phi P_n = 11800^{\text{mm}^2} \times 52.8^{\text{MPa}} = 622.5 \text{ kN}$$



EXAMPLE

Design the columns.
Check the interaction Eq.

$$(a) \text{ For } \frac{P_r}{P_c} \geq 0.2 \Rightarrow \frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

$$(b) \text{ For } \frac{P_r}{P_c} < 0.2 \Rightarrow \frac{P_r}{2P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

Calculate P_r .

$$P_r = P_{nt} + B_2 P_{lt}$$

$$P_{nt} = 495 \text{ kN}; P_{lt} = 43.75 \text{ kN}$$

$$B_2 = \frac{1}{1 - \frac{\alpha \sum P_{nt}}{\sum P_{e2}}}$$



EXAMPLE

Design the columns.

$$\sum P_{e2} = 2x \frac{\pi^2 x 200000^{MPa} x 1010 x 10^6 mm^4}{(3.5 x 7500^{mm})^2} = 5786 kN$$

$$\sum P_{nt} = \sum P_u = 440^{kN} + 110^{kN} + 440^{kN} = 990 kN$$

$$B_2 = \frac{1}{1 - \frac{\alpha \sum P_{nt}}{\sum P_{e2}}} = \frac{1}{1 - 1.0 \frac{990^{kN}}{5786^{kN}}} = 1.21$$

$$P_r = P_{nt} + B_2 P_{lt} = 495^{kN} + 1.21 x 43.75^{kN} = 548 kN$$



EXAMPLE

Design the columns.

$$P_r = P_{nt} + B_2 P_{lt} = 495^{kN} + 1.21 \times 43.75^{kN} = 548 \text{ kN}$$

$$\frac{P_r}{P_c} = \frac{P_u}{\phi P_n} = \frac{548^{kN}}{622.5^{kN}} = 0.88$$

This ratio is very big and there is no room for bending moment.

The problem is related to the relatively narrow flanges ($b_f=209$ mm). Lets try W530 \times 150 with a $b_f=312$ mm.

$\Phi M_p = 1290$ kN-m; $A = 19200$ mm²; $I_x = 1010 \times 106$ mm⁴

$I_y = 103 \times 106$ mm⁴; $r_x = 229$ mm; $r_y = 73.2$ mm.

No Sway Mode: $kL_y/r_y = (7500\text{mm}/73.2\text{mm}) = 102$

$\Phi F_{cr} = 137$ Mpa

$\Phi P_n = 137\text{MPa} \times 19200\text{mm}^2 = 2630$ kN

$P_u/\Phi P_n = 495\text{kN}/2630\text{kN} = 0.19$ (Reasonable.)



EXAMPLE

Design the columns.

Sway Mode: Check in-plane.

Inelasticity: $P/A=495\text{kN}/19200\text{mm}^2=25.8\text{ MPa}$; $\tau=1.0$

$$G_T = \frac{\tau \sum \left(\frac{I_x}{L_c} \right)_{W 530 \times 150}}{m \sum \left(\frac{I_x}{L_b} \right)_{W 610 \times 113}} = \frac{1.0 \left(\frac{1010 \times 10^6}{7.5^m} \right)}{\left(\frac{875 \times 10^6}{12.0^m} \right)} = 5.54 \rightarrow K_x = 3.5$$

$$G_B = \text{pinned} = \infty$$

$$\frac{kL_x}{r_x} = \frac{3.5 \times 7500^{\text{mm}}}{229^{\text{mm}}} = 115 \Rightarrow \phi F_{cr} = 111.3 \text{ MPa}$$

$$\phi P_n = \phi F_{cr} A = 111.3^{\text{MPa}} \times 19200^{\text{mm}^2} = 2136 \text{ kN} \Rightarrow \frac{P_u}{\phi P_n} = \frac{2 \times 548^{\text{kN}}}{2 \times 2136^{\text{kN}}} = 0.25$$



EXAMPLE

Design the columns.

Use the larger $P_u/\phi P_n$ ratio: 0.23

$$\frac{\sum P_u}{\sum \phi P_n} = 0.25 > 0.2 \Rightarrow \text{Use Eq. H1-1a}$$

$$(a) \text{ For } \frac{P_r}{P_c} \geq 0.2 \Rightarrow \frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

$$\frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + 0 \right) \leq 1.0$$

$$M_{rx} = M_{ux} = B_1 M_{nt} + B_2 M_{lt}$$



EXAMPLE

Design the columns.

$$M_{rx} = M_{ux} = B_1 M_{nt} + B_2 M_{lt}$$

$M_{nt} = 0 \Rightarrow$ No moment at columns due to gravity loads.

$$B_2 = \frac{1}{1 - \frac{\alpha \sum P_{nt}}{\sum P_{e2}}}$$

$$\sum P_{e2} = 2x \frac{\pi^2 x 200000^{MPa} x 1010 x 10^6^{mm^4}}{(3.5x 7500^{mm})^2} = 5786 \text{ kN}$$

$$\sum P_{nt} = \sum P_u = 440^{kN} + 110^{kN} + 440^{kN} = 990 \text{ kN}$$



EXAMPLE

Design the columns.

$$B_2 = \frac{1}{1 - \frac{\alpha \sum P_{nt}}{\sum P_{e2}}} = \frac{1}{1 - 1.0 \frac{990^{kN}}{5786^{kN}}} = 1.21$$

$$M_{rx} = M_{ux} = B_1 M_{nt} + B_2 M_{lt} = 0 + 1.21 \times 525^{kN-m} = 635.2 \text{ kN-m}$$



EXAMPLE

Design the columns.

Calculate the available bending strength of the “beam-column.”

$$1.2 \times 525 \text{ kN-m} = 635.3 \text{ kN-m}$$



W530 × 150; $\Phi M_p = 1290 \text{ MPa}$

$C_b = 1.67$; $L_b = 7500 \text{ m}$; $\Phi M_r = 924 \text{ kN-m}$

$L_p = 3.1 \text{ m}$; $L_r = 8.37 \text{ m}$

$L_b < L_r$; Therefore check $C_b \Phi M_r$.

$C_b \Phi M_r = 1.67 \times 924 \text{ kN-m} = 1543 \text{ MPa} > \Phi M_p = 1290 \text{ MPa}$

$\Phi M_n = \Phi M_p = 1290 \text{ MPa}$. (The section is compact.)

Calculate P_r .

$$P_r = P_{nt} + B_2 P_{lt} = 495 \text{ kN} + 1.21 \times 43.8 \text{ kN} = 548 \text{ kN}$$



EXAMPLE

Design the columns.

$$(a) \text{ For } \frac{P_r}{P_c} \geq 0.2 \Rightarrow \frac{P_r}{P_c} + \frac{8}{9} \left(\frac{M_{rx}}{M_{cx}} + \frac{M_{ry}}{M_{cy}} \right) \leq 1.0$$

$$\frac{P_r}{P_c} + \frac{8}{9} \frac{M_{rx}}{M_{cx}} = \frac{548^{kN}}{2136_{kN}} + \frac{8}{9} \frac{635.3^{kN-m}}{1290^{kN-m}} = 0.69 < 1.0$$

The difference is too much. Decrease the size of the “beam-column” keeping in mind that $b_f > 200$ mm.

Try W460 × 113 with a $b_f = 280$ mm. This section will do better. Do similar calculations one more time.

Need to check shear in the “beam-column”.

Also need to check Beam (B) for increased 2nd order moments.

